

REPORT DOCUMENTATION PAGE

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Air Force Research Laboratory (AFMC)

AFRL/PRS

5 Pollux Drive

Edwards AFB CA 93524-7048

8. PERFORMING ORGANIZATION
REPORT

9. SPONSORING / MONITORING AGENCY NAME(S) AND ADDRESS(ES)

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AFRL/PRS

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MEMORANDUM FOR PRS (In-House/Contractor Publication)

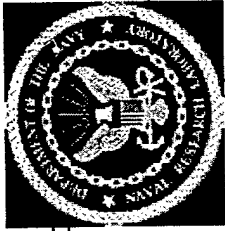
FROM: PROI (TI) (STINFO)

07 Aug 2000

SUBJECT: Authorization for Release of Technical Information, Control Number: **AFRL-PR-ED-TP-2000-162**
I. J. Wysong (AFRL/PRSA); D. C. Wadsworth, D. B. VanGilder (ERC); C. Kaplan, D. Mott
(NRL/LCPFD), "SUPREM-DSMC (CHSSI CFD-8) Software Acceptance Test Review (Draft)"

HPCMO CHSSI Software Acceptance Test Review
(Washington DC, 11 Aug 00) (Submission Deadline: 10 Aug 00)

(Statement A)



SUPREM-DSMC (CHSSI CFD-8)

Software Acceptance Test Review

Agenda:

- Introduction and Overview: I. Wysong
- Software Design: D. Wadsworth
- Grid and Geometry: C. Kaplan
- Particle Movement: C. Kaplan
- Gas/Chemistry: D. Wadsworth
- Boundary Conditions: D. Wadsworth
- Status and Conclusions: I. Wysong

HPCMO/NRL
Washington, D.C.

11 August, 2000

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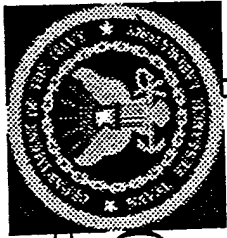
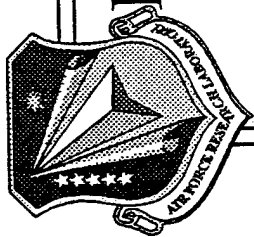
Computational Fluid Dynamics CTA
(Computational Technology Area)
CTA Leader: Dr. Jay Boris

CFD-8: SUPREM DSMC

Scalable, Parallel, Reacting, Multidimensional Direct
Simulation Monte Carlo Flow Code

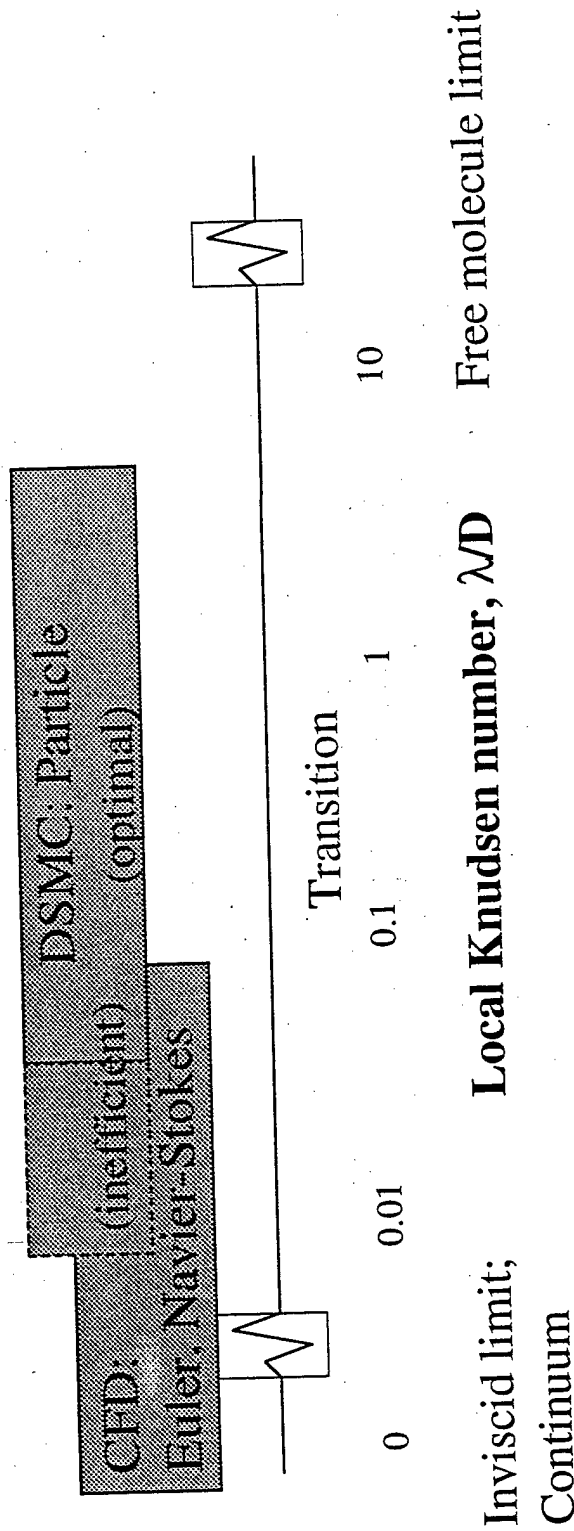
Awarded: November, 1999

Start Date: January, 2000

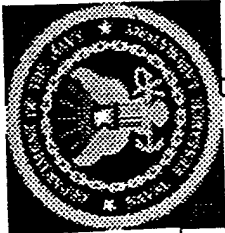
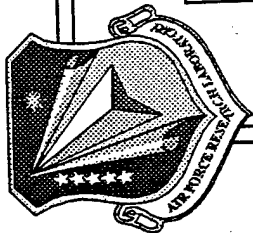


Direct Simulation Monte Carlo method (DSMC)

- Reduced number of model particles represent real gas
- Moved in a grid covering physical space
- Inter-molecular collisions selected statistically from particles in grid cell; physics and chemistry of collisions modeled directly (but phenomenologically)
- No assumption of equilibrium (e.g. N-S eqns.)



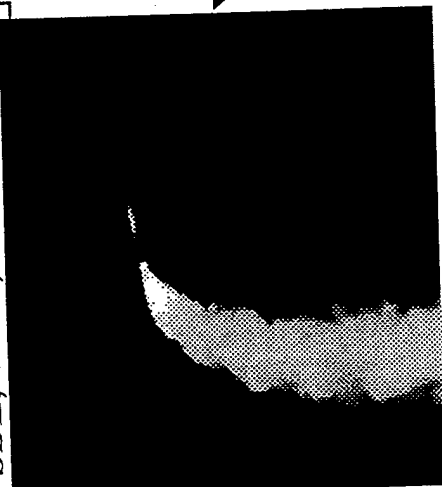
Based on G.A. Bird, "Molecular Gas Dynamics and the Direct Simulation of Gas Flows"



DoD Need for DSMC Simulation Capability:



BattleSpace Environment
Simulation codes:
NTW, THAAD, NMD,
SBL, ABL, SBIRS

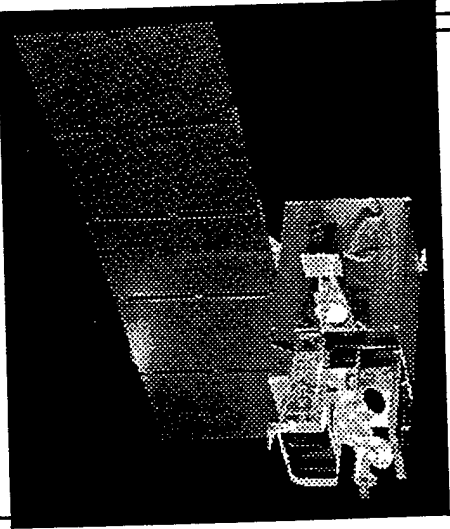


Plasma Processing (etching,
deposition, wafer yield):
Intel, Watkins-Johnson

Micropropulsion, MEMS
(microfluidics):
AF, NASA (JPL), DARPA



Spacecraft Contamination
(lifetime on orbit):
AF (SBIRS, SBL), NASA



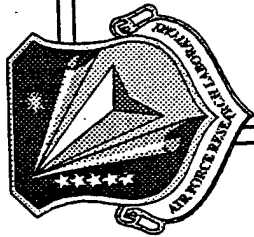
Hypersonic Re-entry Vehicles

(heat transfer, drag...):

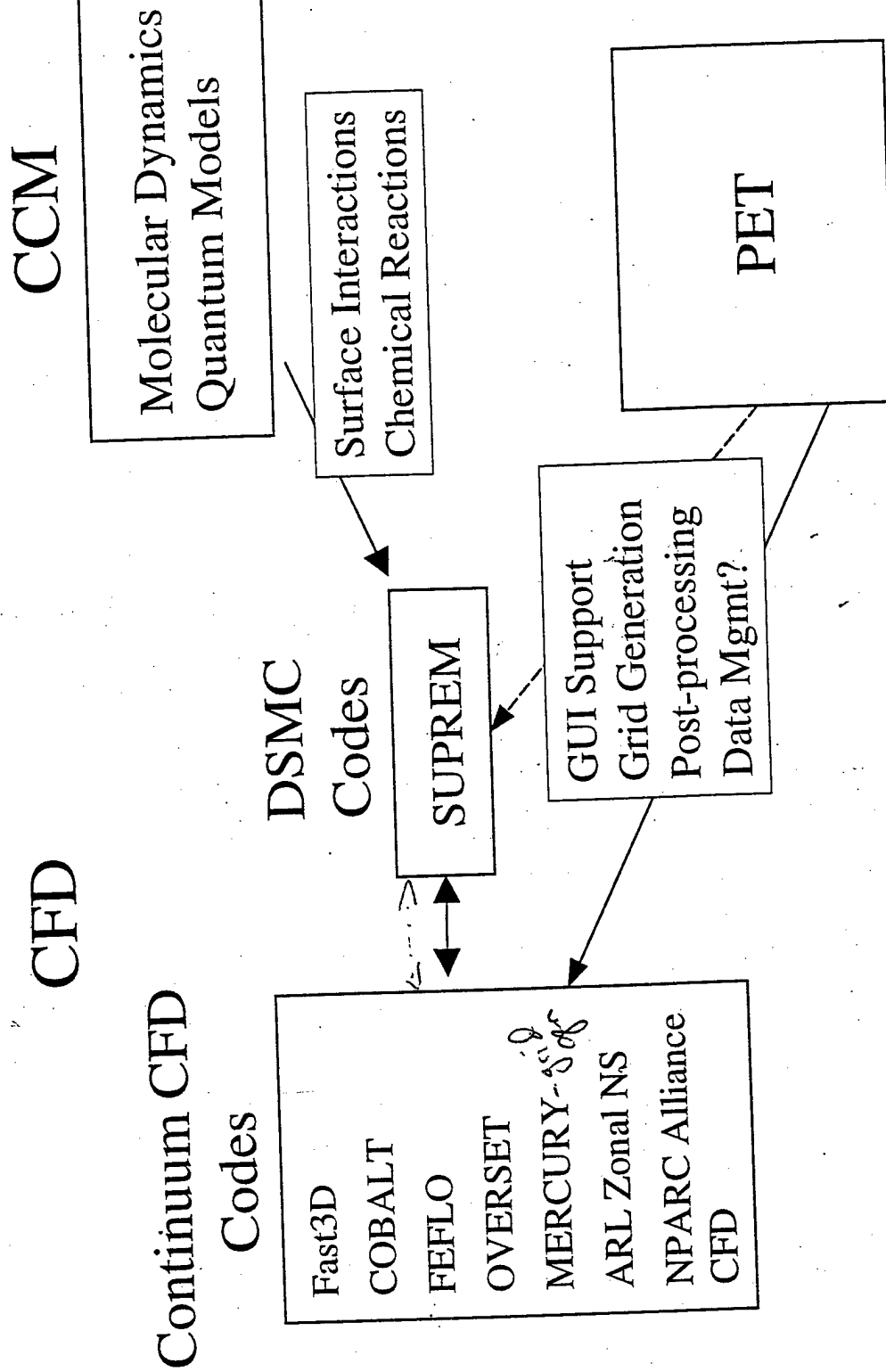
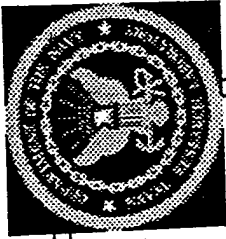
NASA (planetary probes, shuttle),
AF(HyTech, Military Aerospace
Vehicle)

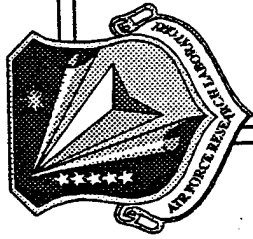
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Integration into HPCMO CHSSI Program





SUPREM Team:

AFRL/PRSA (Edwards AFB, CA):

Dr. Ingrid Wysong
Dr. Dean Wadsworth
Dr. Douglas VanGilder
Dr. David Campbell

NRL/LCP (Washington, DC):

Dr. Carolyn Kaplan
Dr. Elaine Oran
Dr. David Mott

Grid

Code architecture

User Interface, automation

Particle collisions

Boundary Conditions, Sampling

Chemical reaction & species
databases

Geometry

Particle movement through grid

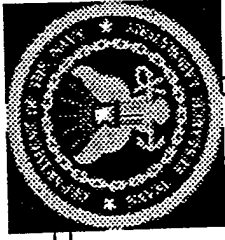
Parallelization

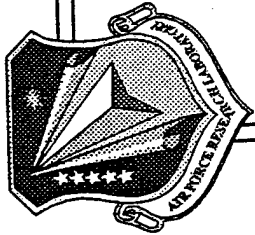
Consultants: International Team of DSMC Experts:

Prof. Graeme Bird (Sydney, Australia)

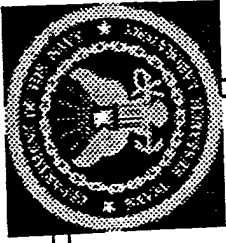
Prof. Mikhail Ivanov (Inst. Theor. Appl. Mech., Novosibirsk, Russia)

Prof. Iain Boyd (Univ. Michigan)

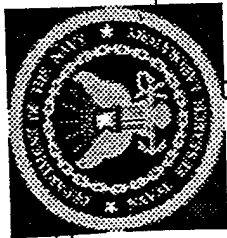
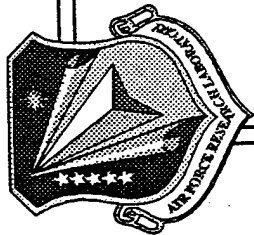




SUPREM Team: Project Management



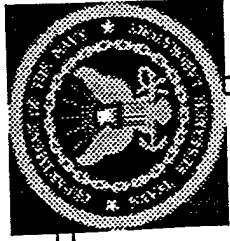
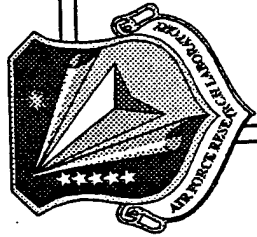
- Geographically distributed:
 - inter-service team brings greater range of experience and breadth of DoD resources.
 - Real-time interaction limited: face-to-face meetings ^{remove extra space} infrequent. Thus, use of email, video and telephone conferences important.
- Software configuration management:
 - Problems:
 - Not a mature process (No HPC MSRC Provision)
 - Single vs. multiple distributed repositories
 - DSMC is "toolbox", i.e., complex input, many models
 - Solution: Aegis (Gnu)
 - 1 version of code
 - changes undergo development, build, test, review, integration
 - each site has repository, synchronized weekly



DSMC Code Availability

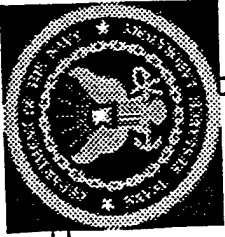
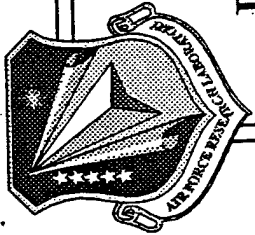
- Number of research & specialized application codes exist.
- None optimized for DoD RDT&E community.

Code: Funding	Ser/ Par.	Geom.	Grid	Lang.	User Interface	Documen tation
Socrates: AF/BMDO	S	3D	Cartesian Static	F77	Text/ Windows	User, Theory
Bird: COTS	S	2D/3D	Cartesian Adaptive	F77	Windows	(user), (theory)
SMILE:	P	2D/3D	Cartesian Adaptive	F77/C	Text/GUI	(user), (theory)
Monaco: Univ.	P	2D/3D	Unstruct. Static	F77/C	Text files	(user)
DAC: NASA	P	3D	Cartesian Static	F77	Text	-
SUPREM: HPCMO	P	3D/2D	Cartesian Adaptive	F90	Text/GUI	User, Theory, & Prog.



SUPREM: Capability Assessment/Feedback

- Detailed DoD User Requirements:
Questionnaire emailed to users, potential users and
DSMC researchers
- Direct Discussions and Input from Users:
AIAA Reno, Jan 00; JANNNAF, May 00; AIAA Denver
June 00
- Direct Discussions with consultants:
AIAA, Reno, Jan 00; AIAA Denver, June 00



SUPREM Approach

Primary requirements: Robust, reliable performance and results for non-expert users (efficiency secondary)

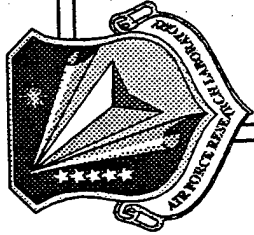
- HPC

- Language: Fortran 90 + MPI [possibly HPF-2]
 - dynamic memory & portable
- Platforms: Initial version to be demonstrated on SGI-Origin, IBM-SP. Then portable to other platforms.

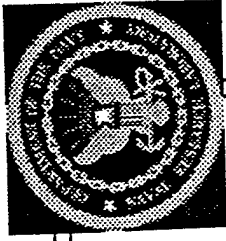
- Software

- Structured, Modular, Modern code and coding practices to enable flexibility and upgradability (supplemented by documentation)
- Hierarchical design:
 - input/interface
 - data structures

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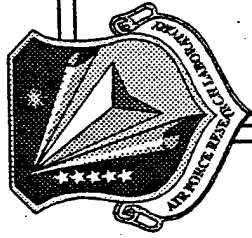


SUPREM: Risk Mitigation



- Build on DSMC core modules and experience of AFRL and NRL teams.
- Input from expert advisory panel.
- Use thoroughly established and tested algorithms.
- Validation of results against established data sets, analytical cases and extent research codes.
- Frequent input from DoD users to ensure product will meet requirements_Q (Tech. transitions/ leveraging)Ⓢ

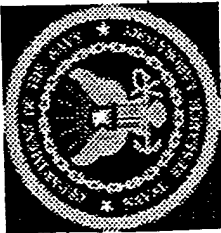
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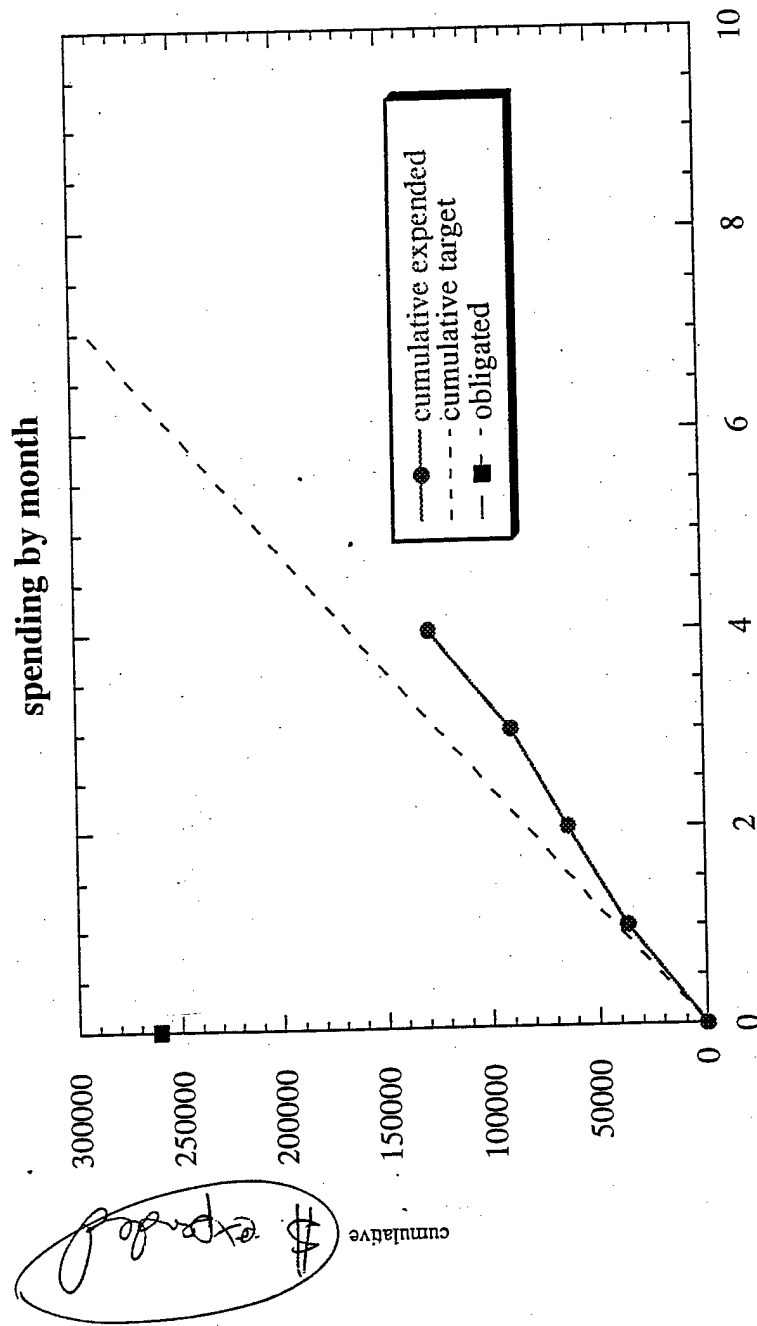
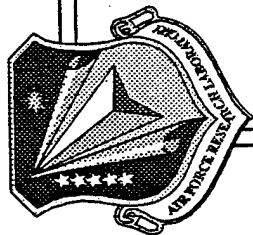
SUPREM: Leveraging

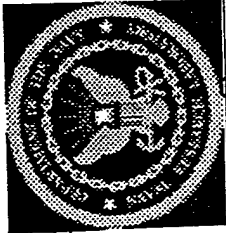


- BMDO -- plume simulation expertise, potential additional funding for radiation module
- NRL DARPA-funded MEMS-flow computations
- AFRL AFOSR-funded research into DSMC collision models and validation cases
- Spectral Sciences SBIR project: comparison case runs for molecular beam simulation, unsteady flowfield computational techniques



CFD-8 Spending Profile

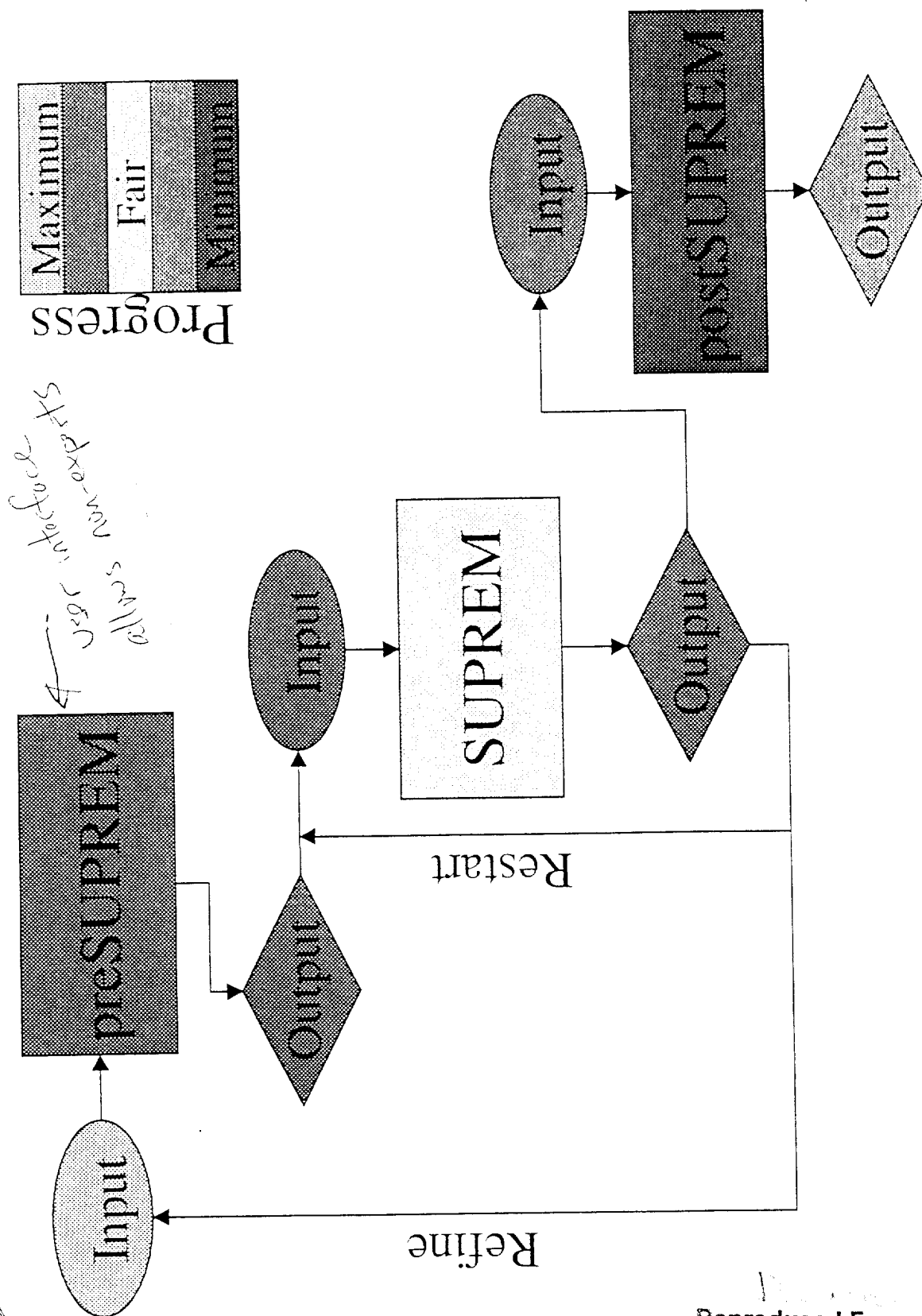
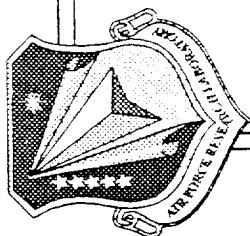




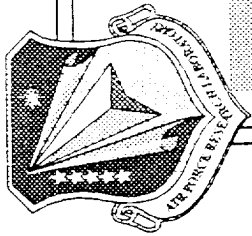
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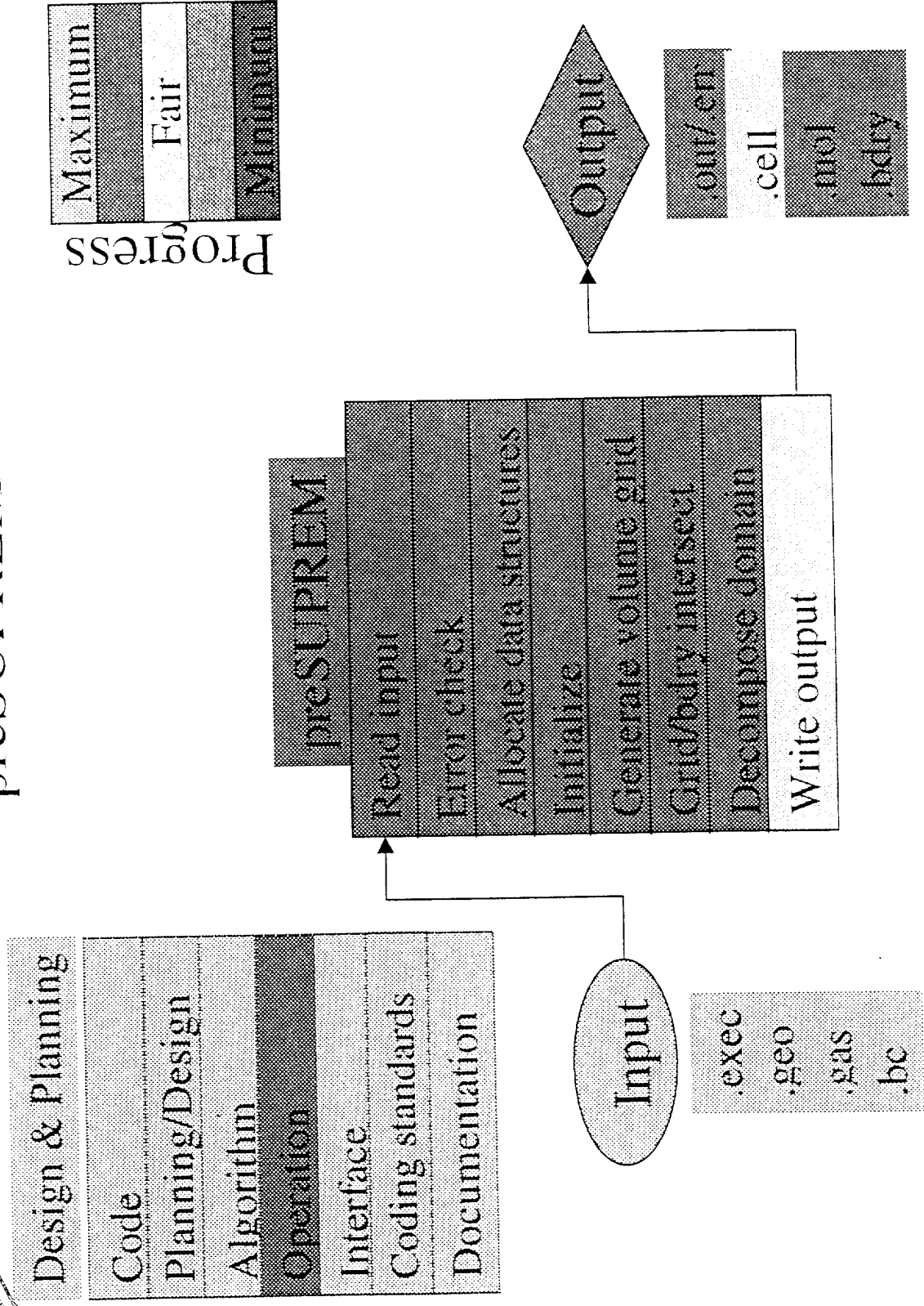
SUPREM-DSMC Overview & Status

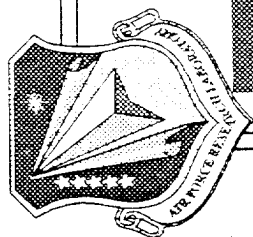


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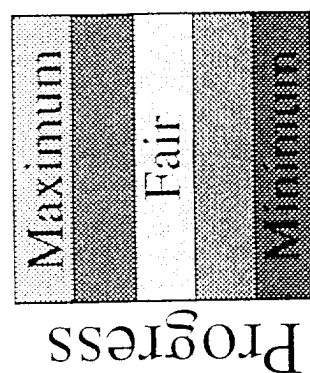
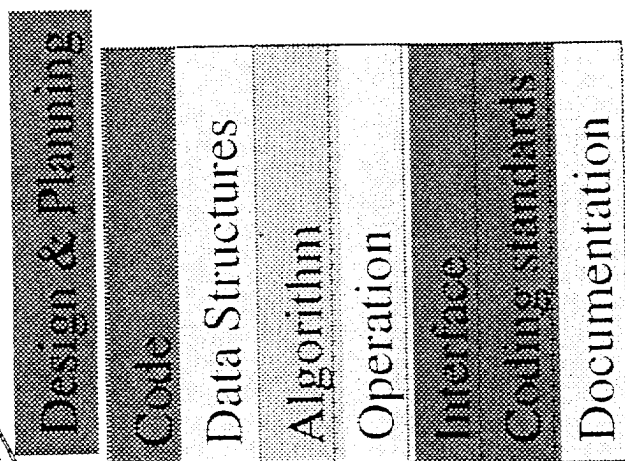


Code Overview & Status: preSUPREM

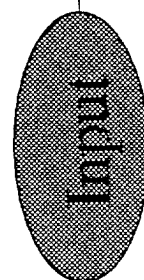
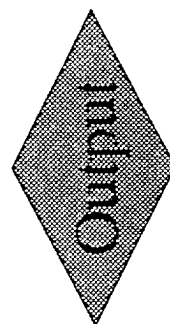
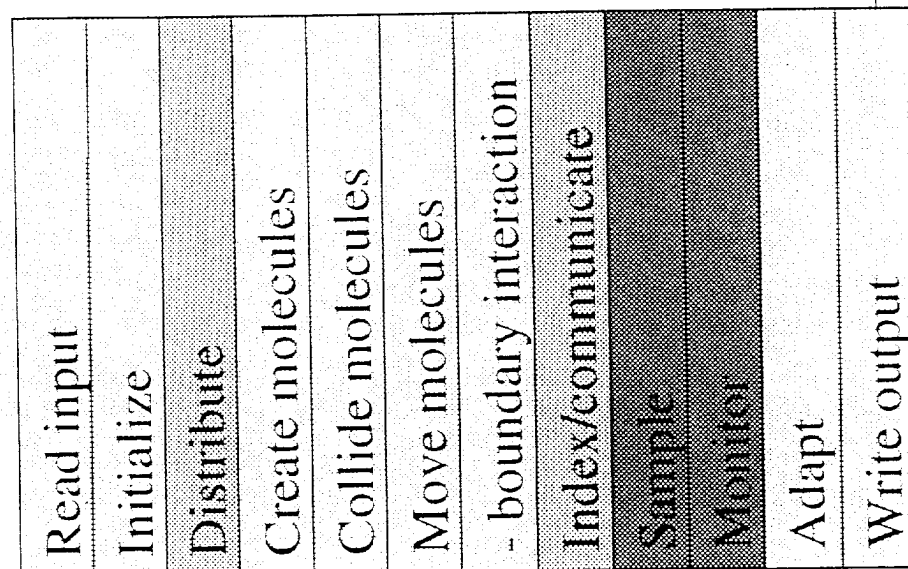




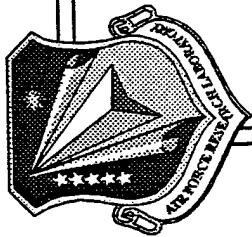
Code Overview & Status: SUPREM



SUPREM

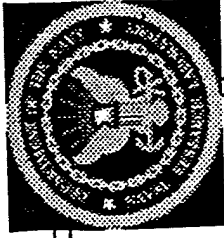


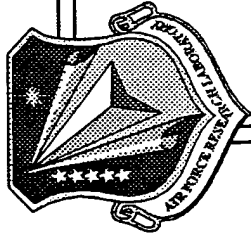
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Software Design

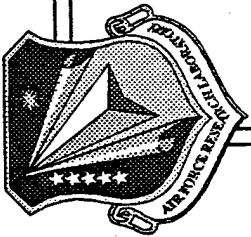
- Configuration Management
- User Interface / Code Operation
- Automation / Adaption



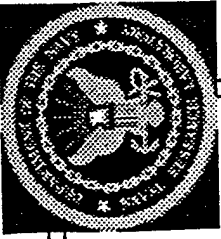


Configuration Management

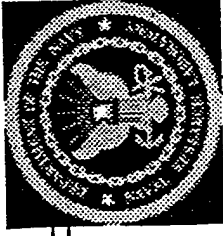
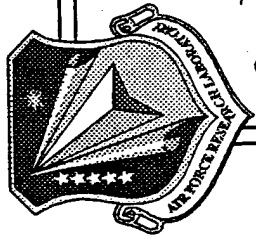
- One Active, Validated Version of Code (Baseline)
- Configuration Management Process:
 - At Minimum: Source Code Version Control
 - Develop / Modify / Build / Test / Review ...
 - Integrate Changes into Baseline
 - Distribute
 - Track Bugs / Fixes
- Desire to Leverage HPC/CHSSI/PET in this Area
 - No CHSSI/PET/MSRC Process Identified
 - MSRC Serves as Repository of Application Codes (Executables)
 - Need “Standards” - Repository of Software Development / Maintenance Expertise



Configuration Management (cont.)

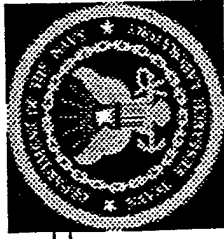
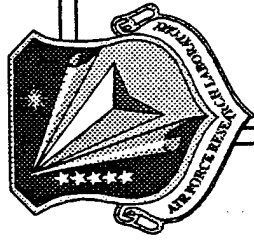


- Aegis (GNU) Selected as C.M. Tool (and C. M. Process)
 - Multiple Architecture Support
 - Full Regression Testing is Integral
- Multiple Local Repositories (ARFL, NRL)
- Changes “Pushed” Weekly
- Repository Includes (& Aegis *Controls*):
 - Source code, etc,
 - Documentation
 - Databases (Gas, Chemistry, etc.)
 - Test, Demonstration, Validation Cases & Results



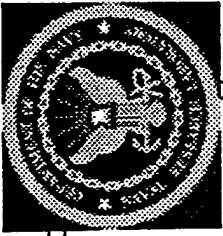
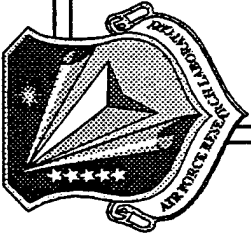
Coding Standards

- Modularity / Extensibility of Source Code and Data Structures is Key Objective *check spacing between lines*
 - Object-Oriented
 - Data-Hiding
 - Data-Proximity
- Exploit Fortran90 [HPF-2?] Features
 - Modules
 - Interfaces
 - Data Structures (TYPE)
 - Parallel Portability / Maintainability (HPF-2)
- Establish Templates for Key Modules / Routines
 - Standardization
 - Re-use
- Data Management?
 - Database Format (HDF/CGNS?) / Interface to External Tools



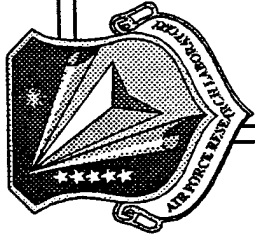
User Interface

- DSMC as Toolbox
 - Grows in Complexity with Time (“Feature Creep”)
 - No Single or Optimal Approach / Algorithm / Model
 - Robustness Makes Maintenance / Control Difficult
- Provide / Enforce Standardized Interface
 - Scalable Physics - Models Classed by Complexity
 - Restrict Features Based on User Expertise
 - Very Complex Input - Automate or Guide Problem Setup
- Control Size / Contents / Usage of Toolbox



User Interface (cont.)

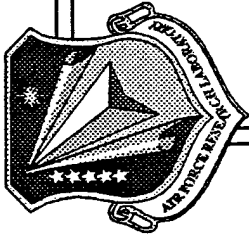
- Hierarchical / Extendable Input
- User Expertise versus Code Features
 - Novice: Restricted Access, Limited Control (Automatic, Conservative Defaults)
 - Intermediate
 - Expert: Full Access, Full Control
- “Problem Type” versus Setup and Model Availability
 - Dominant Flowfield Characteristics (Shock Layer, etc.) can be used to Guide Selection of Models (Gas, Chemistry, Boundary Conditions, Grid, etc.)



Automation

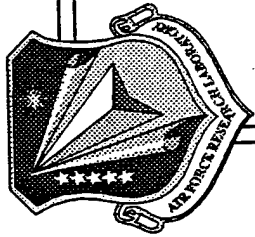
- User Defines Problem, but Code Should Control How Solution is Obtained
- Are the Results Accurate?
 - Satisfy DSMC Algorithm
 - Time Step, Cell Size, Nearest-Neighbor Collision Partners
 - Satisfy Physics of Problem
 - Gas Representation, Chemistry, Boundary Conditions
- Is the Simulation Efficient?
 - Memory, CPU, Load Balance
- Primary Objective: Accuracy, Generality, Scalability, and Robustness of Physics and Algorithm





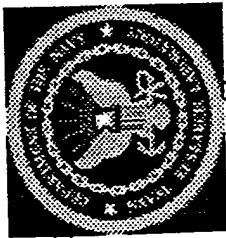
Automation (cont.)

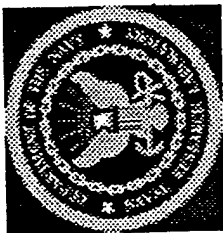
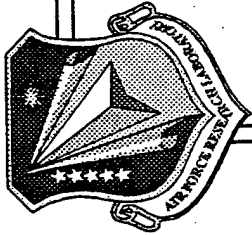
- Very Limited Existing DSMC Work in this Area - Far Short of Conventional CFD
- Plan:
 - Quality / Accuracy Constraints Represented via Metrics
 - Global (Average)
 - Local (Worst Case)
 - Expert User (Only) Can Tradeoff Quality for Efficiency
 - Interpretation of Results Facilitated by Feedback & Output
 - Desired vs. Achieved Quality and Accuracy Metrics
 - Refinement / Adaptation Results



Gas / Chemistry Description

- Representation
- Models
- Algorithms
- Data Structures
- User Access
 - Input
 - Initialization
 - Sampling/Output

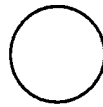




Gas Model

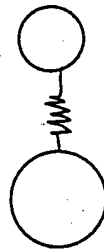
- Multiple Species
- Arbitrary Complexity of Each (Scalable)
 - Number of Independent Internal Modes
 - Representation of Mode (Continuous or Quantized)
 - Each Mode is a Scalar Quantity Stored for Each Molecule

Atom



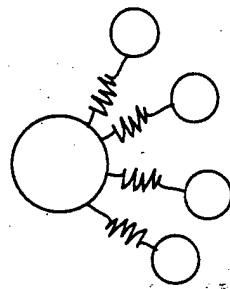
Ar (ϵ_t)

Diatom

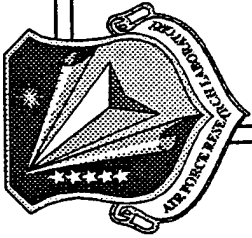


NO ($\epsilon_t, \epsilon_r, \epsilon_v$)

Polyatom

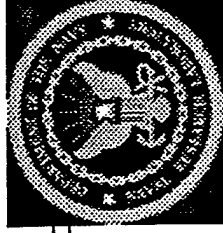
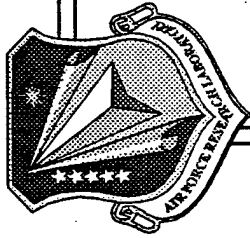


CH₄ ($\epsilon_t, \epsilon_{r1} \dots \epsilon_{rn}, \epsilon_{v1} \dots \epsilon_{vn}$)



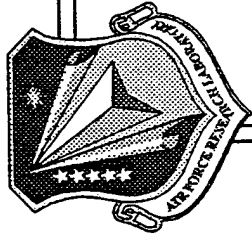
Gas Model

- Predefined Common Modes are Available
 - Rotation -- Rigid Rotor: Quantized, Bounded
 - $\epsilon_j = k_B \theta_j(j+1)$ gives ϵ_r for level j
 - Vibration -- Simple Harmonic Oscillator: Quantized, Bounded
 - Anharmonic Oscillator: Quantized, Bounded
- Allow Fictitious or Simplified Representation
 - Lumped -- Combine (Lump) "Real" Modes Together
 - Arbitrary Subset of Levels in a Quantized Mode



Databases Provided for Common Applications

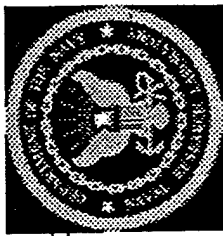
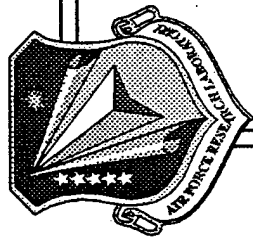
- Completeness/Complexity Parameterized by Flow Energy Content (e.g. Temperature, Velocity, Average Collision Energy) Relative to Characteristic Mode Energy
- Simple Monatomics
 - e.g., Ar
- Common laboratory species
 - N₂
 - Room Temperature / Thermal
 - Weak or Strong Shock Layers
- Air
 - N₂, O₂
 - N₂, O₂, O, N, NO
- Propellents, Combustion Products
 - CO, CO₂, CH, H₂O



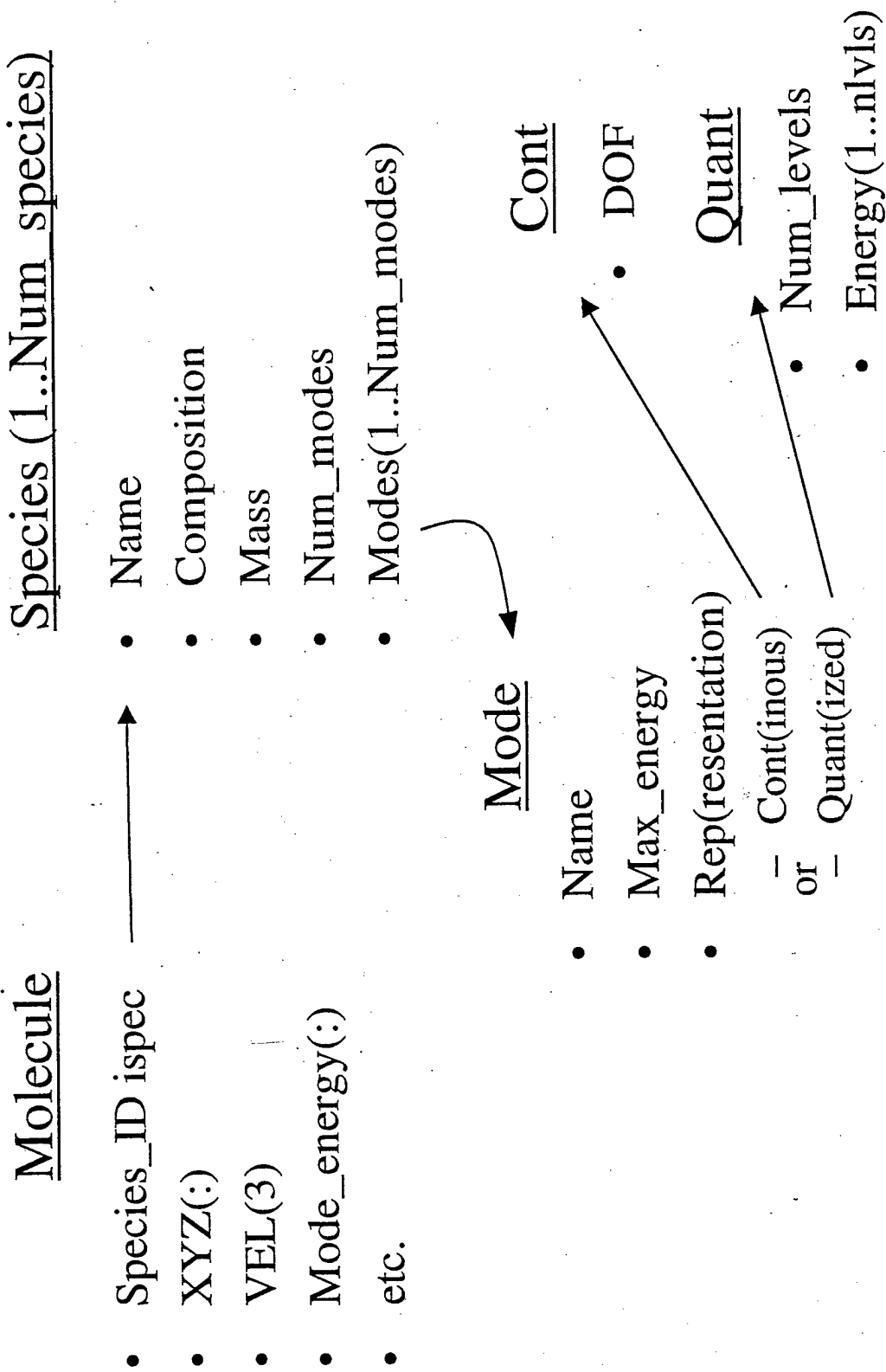
Access to Molecules / Species

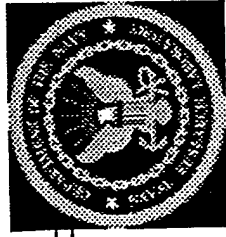
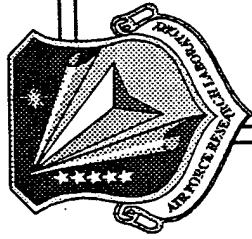


- Input: Compact, Self-contained Definition Section; 1 for Each Species
- Initialization: Molecule of a Given Species can be Assigned an Arbitrary Velocity & Energy in any mode
 - Sampled from Equilibrium (e.g., Temperature)
 - Sampled from Nonequilibrium Distribution (e.g., Discrete Mode Energy Population)
- Sampling/ Output:
 - Default: Mean Properties per Mode in ALL Cells
 - “Detail”: Distribution Function of Any Mode of Any Species (at Any Location) -- Vital for “Microscopic” Interpretation



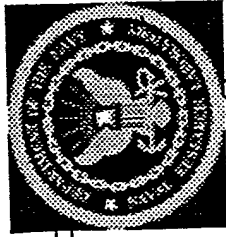
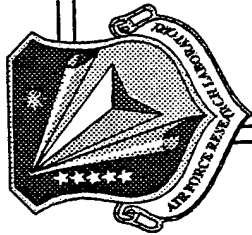
Molecule and Species Data Structures





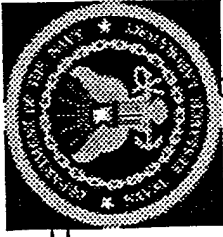
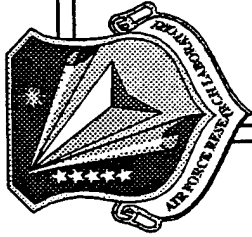
Chemistry (Intermolecular Collision) Model

- Bimolecular [Trimolecular & Unimolecular also Available]
- Arbitrary Complexity of Collision for Each Species Pair (Scalable)
 - Number of Independent Processes
 - Complexity of Each Process
- Collision Description is INPUT (Self-Documenting)
 - 1) Chemical Equation:
 - Defines Species and Modes that Participate
 - 2) Process Keyword Description:
 - Type (Elastic, Inelastic, Reaction)
 - Model Used to Calculate Cross Section σ and Model Parameters



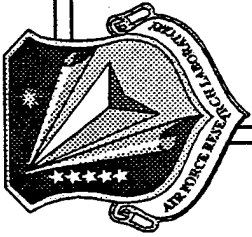
Chemical/ Collision Equations

- $NO + O \rightarrow NO + O$
 - PROCESS: ELASTIC
 - VHS, VSS Model Used to Calculate $\sigma_e = \sigma_e(\mathcal{E}_t)$
 - Isotropic Scattering
- $NO(\mathcal{E}_r) + O \rightarrow NO(\mathcal{E}_r') + O$
 - PROCESS: INELASTIC
 - Collision Number “Z” Model $\sigma_i = \sigma_i(\mathcal{E}_t) \approx \frac{1}{Z} \sigma_e(\mathcal{E}_t)$
 - Borgnakke-Larsen Redistribution (\mathcal{E}_r')
- $NO(\mathcal{E}_r, \mathcal{E}_v) + O \rightarrow N + O_2(\mathcal{E}_r', \mathcal{E}_v')$
 - PROCESS: REACTION
 - Total Collision Energy (Arrhenius-based) Model
 - $\sigma_r = \sigma_r(\mathcal{E}_c = \mathcal{E}_t + \mathcal{E}_r + \mathcal{E}_v)$
 - Borgnakke-Larsen Redistribution/Initialization ($\mathcal{E}_r', \mathcal{E}_v'$)



Collision Representation - Issues

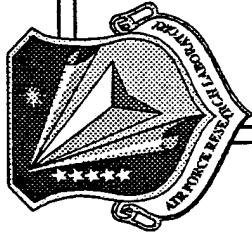
- DSMC Collision Models are Wholly Phenomenological
 - Microscopic/ Kinetic Description Required, but is Nominally Based on Macroscopic Data
 - No Real Cross Sections are Used
- Complex, Real Coupled Processes are Necessarily Decoupled
- Statistical Process: Detailed Balance Achieved by Uncorrelating Pre- & Post- States
- Continued Research into Collision Physics & DSMC-Representation Required
 - SUPREM team
 - CCM



Collision Algorithm - Implementation

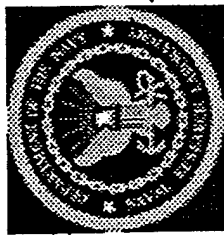
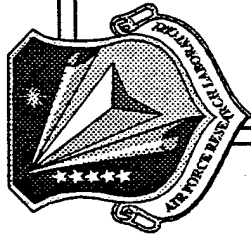
- Pair Selection is Cell-Local
 - Nearest-Neighbor Approximated via Bird Transient Adaptive Subcell Procedure
- Pair & Pair-Process Acceptance
 - Independent of Process Type or Input Order
 - Depends ONLY on (Abstract) Physical Parameter σ
- Null Collision Method
 - Probability of Process

$$P_i = \frac{\sigma_i}{N \sum_{j=1} \sigma_j}$$



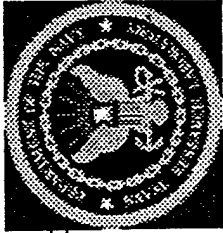
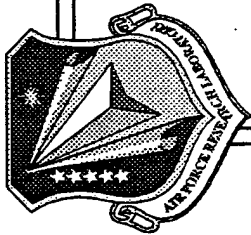
Collision Description - User Issues

- Databases Provided for Common Gases
- User (NOT Black-Box Code) Determines Complexity of Gas & Collision Processes
- Extensive Diagnostics & Feedback from Code
 - DSMC Constraints (e.g., Nearest-Neighbor)
 - Model Realism / Accuracy / Range [Limits] of Applicability



User Access to Collision Processes

- Input: Compact, Self-Documenting for Each Pair / Process
- Initialization: N/A
- Sampling / Output
 - Default: Mean Properties (e.g., Collision Frequencies) Per Pair (& Per Process)
 - Detail: Distribution Function of Any Mode of Any Species (Pre- and/or Post- Collision) -- Vital for Improvement of DSMC Models & Interpretation of Results



Collision Pair Data Structures

Colln_Pair(ispec, ispec)

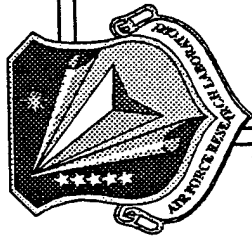
space?

- Weight
- Num_processes
- Process(1..Num_processes)

Process

- Type/name
- Modes_involved
- Model

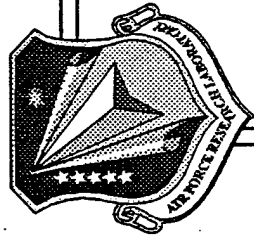
Parameters(1..N)



Gas/ Chemistry Example Input File (Ar-N₂)

1) Species Definition

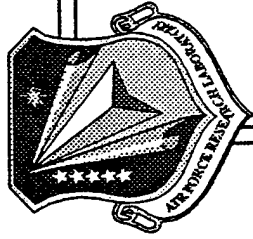
```
SPECIES_NAME Ar
COMPOSITION Ar-1
NUM_INTL_MODES 0
SPECIES_NAME NO
COMPOSITION N-1 O-1
NUM_INTL_MODES 2
MODE_NAME ROT
MODE_REP CONTIN
DOF 2
MAX_ENERGY 1.04E-17
MODE_NAME AHOVIB
MAX_ENERGY 1.04E-17
NUM_DUNHAM_TERMS 3
DUNHAM_COEFFS 0.1 0.2 0.3
```



Gas/ Chemistry Example Input (cont.)

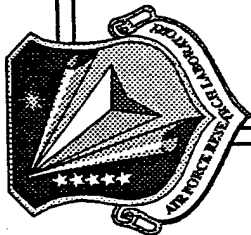
2) Collision Definition

```
BIMOLECULAR TRUE
SPECIES_PAIR Ar ANY
      Ar + ANY -> Ar + ANY
PROCESS ELASTIC MODEL VHS NUM_PARAMS 3
DIAM_REF 4.185e-10
VISC_EXPON 0.80
TEMP_REF 273.
COMMENT ref: Bird 1994, App. A, avg(NO+Ar) "
SPECIES_PAIR NO NO
NO + NO -> NO + NO
PROCESS ELASTIC MODEL VHS NUM_PARAMS 3
...
NO(ROT) + NO -> NO(ROT') + NO
PROCESS INELASTIC MODEL COLLN_NUM NUM_PARAMS 2
COLLN_NUM 5.0
TEMP_REF 300.
SPECIES_PAIR NO Ar
...
```



Boundary Conditions

- Describe:
 - 1 Interaction of Molecule with Boundary During Molecular Movement Stage
 - 2 Spontaneous Creation of Molecules at Boundary (e.g., Inflow)
- Allow Arbitrary Complexity (Scalable)
 - Define Several Basic Types or Classes (General Characteristics)
 - Each Type has Several Subtypes
 - Simple to Complex
 - Highly Constrained to General / Arbitrary
 - Constant or Prescribed Variation in Space or Time

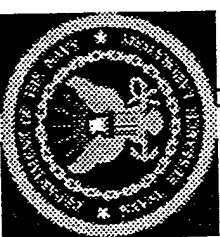
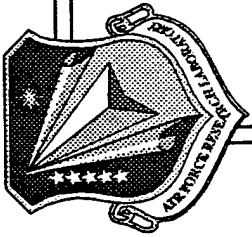


Boundary Conditions

800-513-5000



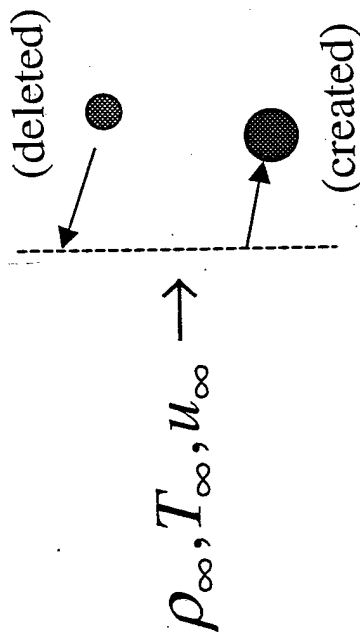
Type	Subtype	Interaction	Creation
SOURCE	MACRO	Delete	Constant, Equilibrium
	MSIS90	Delete	Constant, Equilibrium
	MANUAL	Delete	Arbitrary (f(x,y,z), Noneq.)
	MOL_LIST	Delete	[Externally Supplied]
	ITERATIVE	Delete	Adjustable Rate, Equilibrium
SINK	VACUUM	Delete	N/A
	PUMP	Possible Reflection	N/A
WALL	OUTGAS DESORB	Reflection/Reaction	N/A
		Reflection/Reaction	Constant, Equilibrium
		Reflection/Reaction	Coverage-Dependent, Equilibrium
		Specular Reflection	N/A
SYMMETRY		No Change	N/A
FLUXPLANE	LINEAR AZIMUTHAL	Linear Displacement	[Displ. from matching bdry]
		Rotational Displ.	[Displ. from matching bdry]



X

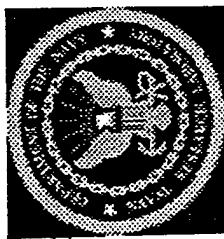
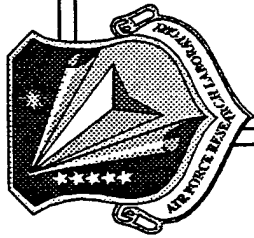
Source Boundary Condition ?

- Typical Case: Constant Inflow, Known Freestream Macroscopic Properties ($n_\infty, p_\infty, T_\infty, u_\infty$)
- General Case: Arbitrary Spatial Variation (e.g., U Profile), Inter-species, Inter-mode, Intra-mode Nonequilibrium



Issues:

- Subsonic vs. Supersonic (Directed Fluxes)
- Macroscopic vs. Microscopic Description



Source / MACRO

- Constant Incoming (Created) Flux
- Incoming Flux Independent of Outgoing Flux
- Strictly Valid Only for High Supersonic Speeds
- Molecule Properties Sampled from Analytical Equilibrium Kinetic Flux

$$\overline{\text{Flux}} = n \overline{Qu_n} = n \int Q u_n f dv; \quad f \propto \exp \left(\frac{-mc^2}{2k_B T} \right)$$

Flux:

$Q =$

Mass

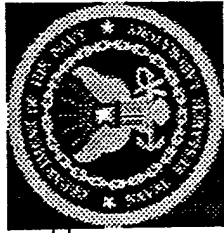
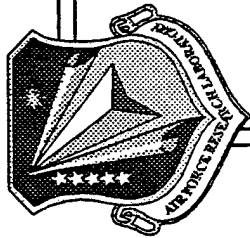
m

Momentum

$m \vec{v}$

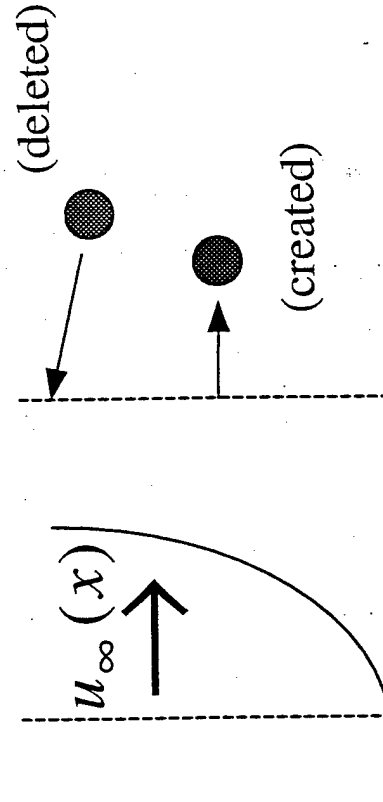
Energy
(Trans.)

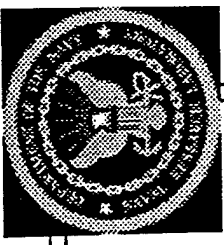
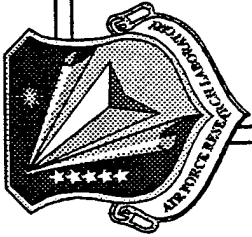
$m |v|^2$



Source / MANUAL

- Generalized Spatial Variation of $\rho_{\infty}, T_{\infty}, u_{\infty}$
- Independent Properties for Each Species
- Independent Properties for Each Mode
 - $T_{\infty \text{ mode}}$ or Distribution/Population
 - (Also $T_{\infty x,y,z}$ or $f(u,v,w)$)
- Incoming Flux Uncoupled from Outgoing



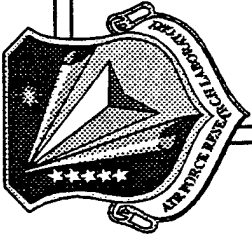


Source / MSIS90

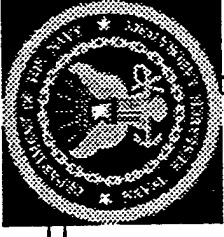
- Flight in Earth Atmosphere
- MSIS90 Model Calculates Freestream Properties from Given Flight Parameters
- Remainder Equivalent to Source / Macro

Source / Iterative

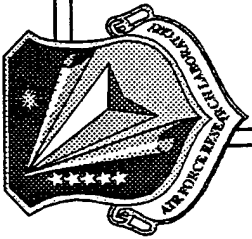
- Couple Incoming & Outgoing Fluxes
- Update Incoming Flux Parameters to Recover Desired Net Flux
- Approximate Subsonic Boundary
- Still Based on Equilibrium, Macroscopic Parameters Q_e



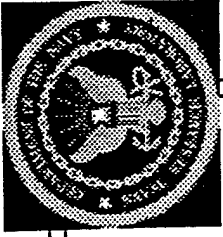
Wall Boundary Condition



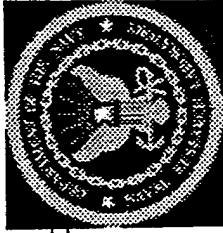
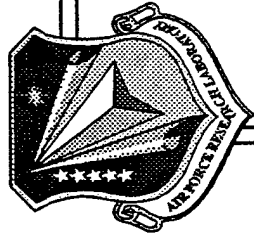
- Typical Case: Constant Temperature Engineering Surface (Diffuse, Fully Accommodating Reflection)
- General Case:
 - $T_w(x,y,z)$
 - Species & Mode-Dependent Reflection & Reaction
 - Spontaneous Creation (Outgassing, Desorption)
- Two Components to Description:
 - Properties Associated with Wall Surface
 - Material
 - Temperature, Velocity, Coverage, Spatial Variation
 - Properties Associated with Interaction
 - Process, Model, Parameters



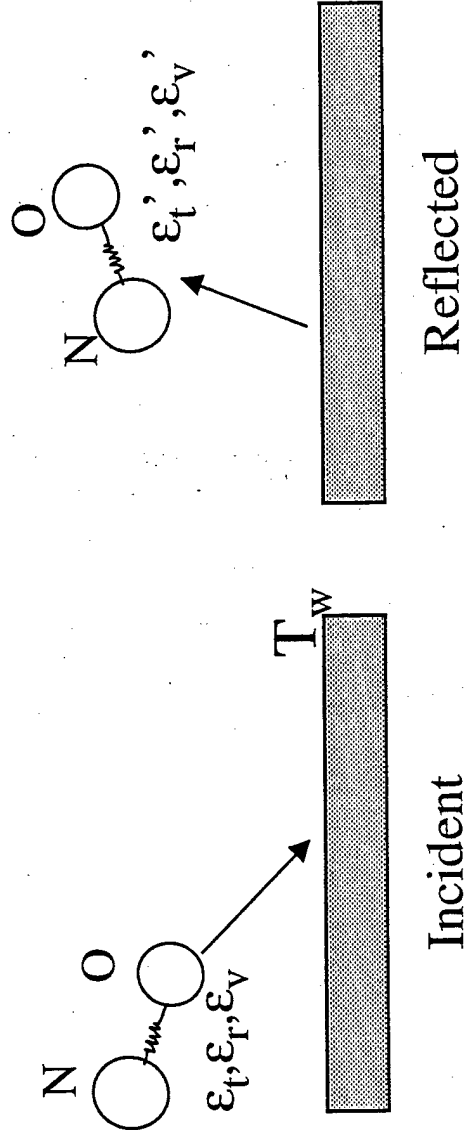
Wall Interaction Processes



- Analogous to Chemistry/Collision Process -
 - Probability (cross section) of Post-interaction State
- Arbitrary Number and Type of Processes for Each Incident Species / Material Pair (Scalable)
- Interaction Description is Input
 - Chemical Equation - Species & Mode
 - Process Definition
 - Type / Model / Model Parameters



Wall Interaction Equations Example - Reflection

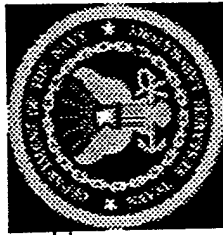
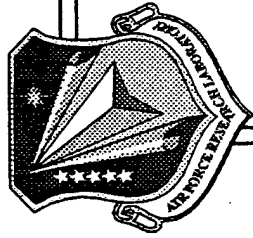


$$NO(\epsilon_t, \epsilon_r, \epsilon_v) + \text{MATL} \rightarrow NO(\epsilon'_t, \epsilon'_r, \epsilon'_v) + \text{MATL}$$

PROCESS RELECTION MODEL MAXWELL NUM_PARAMS 2

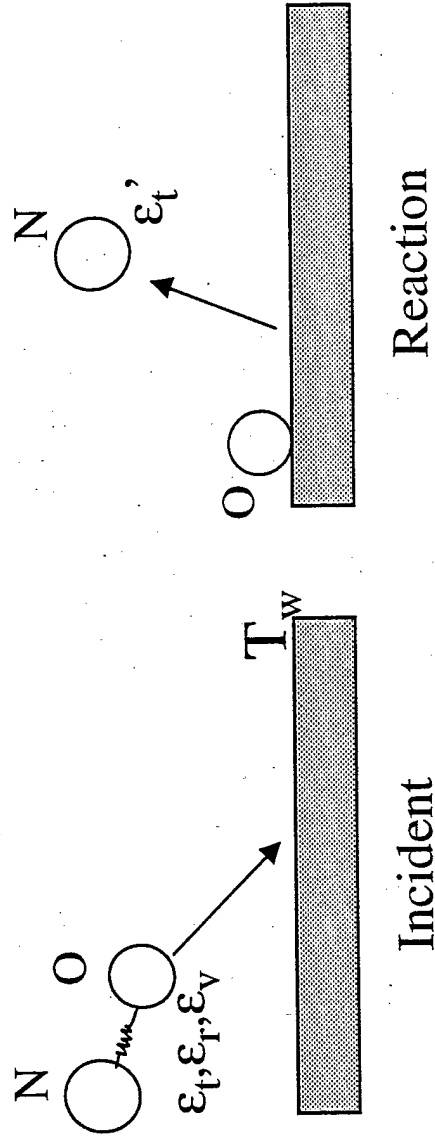
PROB_DIFF 0.5

TEMP_REF 300.0



Wall Interaction Equations

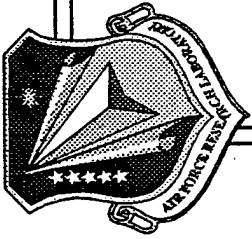
Example - Reaction



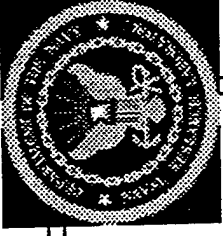
PROCESS REACTION MODEL CATALYTIC NUM_PARAMS 2

PROB_CATALYTIC 0.9

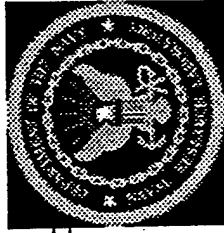
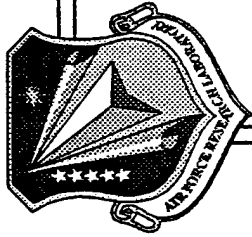
TEMP_REF 1000.0



Wall Interaction Representation Issues



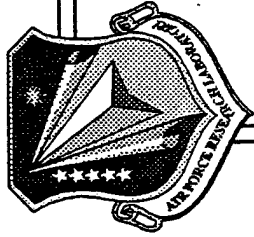
- DSMC Interaction Models are Wholly Phenomenological
 - Microscopic/Kinetic vs. Macroscopic
 - Typically *Independent* of Collision Energy
 - Decoupled from Wall Response
 - Less Sophisticated than Gas-Gas Collision Models
- Much More Research Required
- Present Approach:
 - Implement Available Models
 - Standardize Input & Algorithm
 - Provide Feedback (Accuracy / Realism)



Wall Interaction Algorithm

- Movement Routine Provides:
 - Incident Molecule
 - Boundary Element Involved
- Combination of Serial & Null Collision Schemes
 - Event Probability Independent of Input Order
 - Reflection: Serial, Independent for Each Mode
 - Reaction: Null Collision Evaluation of:

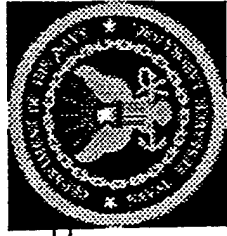
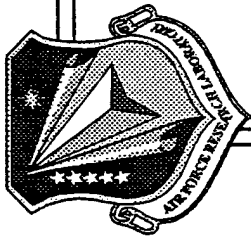
$$\text{Event } i = \frac{p_i}{\sum_{j=1}^N p_j}$$



Wall Interaction Description - User Issues



- Very Limited Data for any but “Fully Accommodating, Diffuse Reflection” Process
- User Defines / Controls Complexity of Interaction
- Extensive Feedback
 - DSMC Constraints (Surface Element Size, Sample Size)
 - Model Realism / Accuracy



User Access to Interaction

- Input: Compact, Self-Documenting For Each Species & Process
- Initialization: N/A
- Sampling / Output:
 - Default: Mean Incident and Reflected Species Properties (Fluxes)
 - Detail: Distribution Function of Any Mode of Any Species (Pre- & Post-interaction)
 - Macroscopic: Surface Pressure, Shear
 - Integrated: Forces & Moments on a Component
- (Long Term:) Couple to Material / Surface Response Model